The precursor structure in relativistic shocks

Barel Skuratovsky,¹* Yuri Lyubarsky,² and Tsvi Piran¹

¹Racah Institute of Physics, The Hebrew University of Jerusalem, Jerusalem 91904, Israel
 ²Physics Department, Ben-Gurion University, Be'er-Sheva 84105, Israel

Accepted XXX. Received YYY; in original form ZZZ

ABSTRACT

We present a common unifying macroscopic framework for precursors in relativistic shock waves. These precursors transfer energy and momentum from the hot downstream to the cold upstream, modifying the shock structure. Derishev & Piran (2016) have shown that in a steady state, there is a maximal fraction of the downstream energy flux that the precursor can carry. We show that at this critical value, the shock disappears, and the flow passes through a sonic point. This behavior resembles the classical Newtonian Rayleigh flow problem. At the critical value, the transition is unstable as perturbations in the upstream accumulate at the sonic point. Thus, if such a point is reached, the shock structure is drastically modified, and the flow becomes turbulent.

Key words: shock waves – relativistic processes

1 INTRODUCTION

Relativistic shock waves are ubiquitous in numerous objects, ranging from AGNs to Gamma-Ray Bursts. A common image of a shock is of an abrupt discontinuity with no information passing from the downstream to the upstream. However, in many cases, this picture is oversimplified. A precursor can propagate ahead of the shock into the upstream. Precursors that can overtake the shock include fast particles (sometimes called cosmic rays), radiation (as in the case of radiationmediated shocks), or supersonic plasma waves. If sufficiently strong, they can significantly modify the upstream and, consequently, the overall shock structure.

The role of fast particles as precursors of Newtonian shocks has been extensively studied since the classical work of Bell (2004). It was later extended to relativistic shocks (e.g. Couch et al. 2008; Nakar et al. 2011; Lemoine et al. 2019). In radiation-mediated shocks, photons transfer energy from the downstream to the upstream. There, the shock structure is completely dominated by this process, and the width of the region is the mean free path of the photons. Detailed studies of such shocks have been carried out both numerically and analytically (Budnik et al. 2010b; Nakar & Sari 2012; Granot et al. 2018). A different precursor was considered within the "pairbalance" model (Derishev & Piran 2016). In this model, high-energy photons generated in the downstream annihilate with low-energy ones in the upstream near the shock, transferring energy and momentum to the upstream and modifying its structure. Garasev & Derishev (2016) have shown using numerical simulations that this may lead to a Weibel instability and generation of magnetic fields upstream. Another modification of the upstream structure can arise from low-frequency electromagnetic waves emitted by magnetized relativistic shocks (Lyubarsky 2018). In particular, such an interaction in electron-ion flows leads to electron heating and even to the non-thermal particle acceleration (Lyubarsky 2006; Hoshino 2008; Sironi & Spitkovsky 2011; Iwamoto et al. 2022).

All these phenomena incorporate energy and momentum transfer from the downstream to the upstream. Often, these fluxes are negligible, but at times, as in radiation-mediated shocks and in the "pair balance" model, they are dominant. We discuss here a general framework for the macroscopic modification of a shock wave due to energy and momentum transfer from the shock's downstream to the upstream. In §2 we describe, following Derishev & Piran (2016), the steady state problem for a relativistic shock. There is a maximal fraction of the total energy that can be transferred from the downstream to the upstream in a steady-state solution. This result resembles the Newtonian Rayleigh flow problem, Concerned with a flow modified by heat deposition (or removal), in which there is also a maximal amount of heat that can be added (or removed) from the flow. At the maximum, the shock wave disappears completely, and the solution passes through a sonic point. In §3 we show that as the flow velocity approaches the speed of sound (from above), instability develops. Possible implications of our results to different astrophysical models and, in particular, of the turbulent flow near the critical point for magnetic field build-up and particle acceleration are discussed at 4.

2 STEADY STATE SOLUTION

We consider the energy and momentum transfer from the shock downstream to the upstream. The process is studied in the shock frame. The overall structure of the flow is the following (see Fig. 1). The shock is placed at $x^1 = 0$, and the flow is directed towards negative x. At the far upstream (formally at $x^1 \rightarrow \infty$) the flow is cold. The relativistically hot plasma downstream emits a fraction of its energy upstream in the form of radiation or relativistic particles. This energy is absorbed in an extended region upstream of the shock so that the flow is heated and decelerated. As a result, the shock discontinuity decreases and may even disappear at all. Global energy conservation implies that far downstream (formally at $x^1 \rightarrow -\infty$), the flow parameters are related to the far upstream parameters by the

^{*} E-mail:barel.skuratovsky@gmail.com

2 B. Skuratovsky et al.



Figure 1. A schematic description of the energy S^0 and momentum S^1 fluxes from the downstream (where the extraction region is marked in red) to a point in the upstream (marked in green). Generally, the energy and momentum deposition is not limited to a single location, and the fluxes are a function of the distance from the shock.

standard Taub adiabat. We are now going to study how absorption affects the parameters of the upstream flow and of the shock wave.

The relativistic flow of a perfect fluid is described by an energymomentum tensor

$$T^{\mu\nu} = w u^{\mu} u^{\nu} + g^{\mu\nu} p , \qquad (1)$$

where *p* is the pressure, $w \equiv e + p$ is the enthalpy, *e* is the energy density, u^{γ} is the four-velocity, $g^{\mu\gamma} \equiv \eta^{\mu\gamma}$ is the flat space metric. The motion is one-dimensional and stationary in the rest frame of the shock. In this case $u^{\mu}(x) = (\Gamma(x), \Gamma(x)\beta(x), 0, 0)$, where we denote by $x \equiv x^1$ the spatial coordinate along which the shock moves. The transfer of energy and momentum from the downstream to the upstream is described by the vector function $S^{\mu}(x) \equiv (S^0(x), S^1(x), 0, 0)$, the components of which are defined as the energy and momentum fluxes emitted from the downstream and absorbed by the upstream flow when it moves from infinity to the distance *x* ahead of the shock. Then, the equations of motion are written as:

$$\frac{\partial T^{\mu\nu}}{\partial x^{\nu}} = \frac{dS^{\mu}}{dx} \ . \tag{2}$$

The equations of motion are supplemented by the continuity of the baryon number density:

$$\frac{\partial (nu^{\mu})}{\partial x^{\mu}} = 0.$$
(3)

The absorbed energy flux, S^0 , is conveniently normalized by the total upstream energy density flux:

$$S^{0}(x) \equiv Q(x)T_{u}^{01} , \qquad (4)$$

where the subscript u denotes the far upstream region, where no energy has been absorbed yet. Recall that the energy fluxes far upstream and far downstream are equal. The function Q(x) defines the fraction of the downstream energy flux that reaches a point distanced x from the shock. In this case, $Q(\infty) = 0$ (as we assume that all the escaping flux is absorbed), whereas $Q(0) = Q_0$ is the total fraction of energy that is extracted from the downstream. Q(x) decreases in the upstream as less and less flux reaches further away from the shock. It increases in the downstream as more and more flux is emitted, and it reaches a maximum at the shock. The absorbed momentum, $S^1(x)$, is assumed to be proportional to absorbed energy:

$$S^{1}(x) \equiv \eta \beta_{u} S^{0}(x) , \qquad (5)$$

where $\eta \leq 1$ is a constant. We describe the system as composed of two fluids: Baryon fluid that dominates the mass density and remains cold in the upstream flow, and radiation and lepton fluid that could dominate the energy density and pressure. In addition to the original electrons in the system, pairs can be produced in the absorption



Figure 2. Q vs. $\beta\Gamma$ for different values of η (for $\beta_{\rm u}\Gamma_{\rm u} = 30$). The red line marks the speed of sound for $\eta = 0$, defined at dQ = 0. $\beta_{\star} = 1/\sqrt{3}$ is a good approximate for β for which Q is maximal.

region. The leptons are relativistic, and their equation of state that is combined with the radiation equation of state satisfies:

$$p = \frac{e_l}{3} = \frac{w_l}{4} \,, \tag{6}$$

where we ignore the mass of the leptons. The far upstream flow is cold, such that:

$$w_{\rm u} = n_{\rm u} . \tag{7}$$

We use units in which the proton's rest mass and the speed of light are unity, $m_p = c = 1$. In the absorption region, we have a combination of baryons (whose thermal energy and pressure are neglected), leptons (whose rest mass is neglected), and radiation. The upstream enthalpy is written as

$$w = n + \frac{4}{3}e_l . \tag{8}$$

Under these assumptions, the conservation laws are written as:

$$n\beta\Gamma = n_{\mu}\beta_{\mu}\Gamma_{\mu} , \qquad (9)$$

$$w\beta\Gamma^2 = n_{\rm u}\beta_{\rm u}\Gamma_{\rm u}^2 \left[1 + Q(x)\right] , \qquad (10)$$

$$p + w(\beta\Gamma)^2 = n_{\rm u}(\beta_{\rm u}\Gamma_{\rm u})^2 [1 - \eta Q(x)] . \tag{11}$$

Eliminating from these equations n, p and w, we obtain:

$$Q(x) = \frac{(1 + 4\Gamma_{\mathrm{u}}\beta_{\mathrm{u}}\Gamma\beta)\Gamma - (1 + 4\Gamma^{2}\beta^{2})\Gamma_{\mathrm{u}}}{4\eta\Gamma_{\mathrm{u}}\Gamma^{2}\beta\beta_{\mathrm{u}} + (1 + 4\beta^{2}\Gamma^{2})\Gamma_{\mathrm{u}}} .$$
(12)

This equation relates the four-velocity at any point to the "returning" flux at this point (measured in units of the upstream energy flux), Q(x). Once Γ is known, we determine *n* from the continuity equation and *w* from Eq. (10). Fig. 2, depicts the dependence of Q on $\Gamma\beta$ for $\beta_{\rm u}\Gamma_{\rm u} = 30$ for several values of η (See also Fig. 2 of Derishev & Piran 2016). First, we notice that momentum transfer changes quantitatively but not qualitatively the results. Notably, $Q_{\rm max}$ decreases monotomically as η increases. In fact, For $\Gamma_{u}\beta_{u} = 30$, the maximal value of energy transferred is only ≈ 0.17 . When the energy transfer is maximal, the flow velocity at the location of the shock decreases to the local speed of sound; see below. In this case, the shock disappears completely, and the flow is continuously decelerated. The part of the



Figure 3. Q_{max} as a function of $\beta_{\mu}\Gamma_{\mu}$. The maximal value of Q_{max} is 1/4, achieved when $\beta_{\mu}\Gamma_{\mu} = 2$ and $\eta = 0$.

curve to the right of the maximum describes the deceleration of the upstream from the initial velocity. The part to the left of the maximum describes the deceleration in the downstream until the Taub adiabat determines the velocity.

The value of Q_{max} as a function of η and the upstream conditions is given by:

$$Q_{\max}(\eta, \beta_u) = \frac{Q_{\max}(\eta = 0, \beta_u)}{1 + 4\eta \beta_\star \beta_u / (1 + 3\beta_\star^2)}$$
(13)

 Q_{max} is shown at Fig. 3. One can see that it is maximal at $\Gamma_u \beta_u \approx 2$. If $\Gamma_u \gg 1$, the upstream flow becomes relativistically hot, and the speed of sound approaches $\beta_s = 1/\sqrt{3}$. As $\Gamma_u \to \infty$, the maximum of Eq.(12) is achieved at $\beta = 1/\sqrt{3}$. At this limit,

$$\lim_{\Gamma_u \to \infty} Q_{\max} = \frac{2 - \sqrt{3}}{\sqrt{3} + 2\eta}$$
(14)

For a given $Q_0 < Q_{\text{max}}$, the relevant part of Fig. 2 for the upstream flow is to the right of the maximal value point. The β values corresponding to Q_0 are the velocity jump at the shock. The region of the curve for values of β smaller than the speed of sound corresponds to the downstream regime in which the energy extraction is building up, from zero at the distant downstream up to Q_0 at the shock discontinuity.

As an example of a possible structure of the upstream (and the downstream), Fig. 4 depicts the modified shock profile as a function of x, given an absorption (and emission) profiles

$$Q(x) = Q_0 \cdot \begin{cases} e^x & x < 0, \\ e^{-x} & x > 0. \end{cases}$$
(15)

In the downstream region (x < 0), Q(x) describes the build-up of the flux due to emission in this region. In the upstream region it describes the decreasing absorbed flux. Note that this absorption law is schematic, and generally relativistic effects make the absorption law much more complicated (see e.g. Granot et al. 2018). In spite of its simplicity, the resulting profile resembles those found numerically by Lemoine et al. (2019) in PIC simulations in which high energy particles accelerated in the downstream deposit their energy in the upstream.

Consider now a flow with $Q_0 = Q_{\text{max}}$. At this point, the thermodynamic heat gains a maximum dQ = 0 at the transition from upstream to downstream. The velocity at that point is denoted by β_{\star} . This, in



Figure 4. $\beta(x)\Gamma(x)$ for an absorption law of (15), $\Gamma_{u}\beta u = 30$ and $\eta = 0$.



Figure 5. The velocity at the sonic point as a function of $\Gamma_{u}\beta_{u}$ for different values of η .

turn, implies d(w/n) = (1/n)dp. Using in the last expression the equation of state, (6), and evaluating w and n using Eqs. (12)-(9) we find a relation between Q_{max} and Γ_* :

$$Q_{\max} = \frac{\Gamma_{\star}^3}{\Gamma_u (3 - 2\Gamma_{\star}^2)} - 1.$$
 (16)

Using $\beta_s^2 = \left(\frac{dp}{de}\right)_S = (w - n/3w)_S$ we find

$$\beta_{s}^{2} = 1 - \frac{1}{\Gamma_{\star}^{2}} = \beta_{\star}^{2} .$$
(17)

Thus, the velocity of the flow achieves the local speed of sound at the point where the absorbed energy reaches Q_{max} . Equating Eqs. (12) and (16), we find the equation for $\beta_{\star} = \beta_{\text{s}}$:

$$\Gamma^3_{\star}(\beta^3_{\star} + \eta\beta_u) + \Gamma_u\beta_u(2\Gamma^2_{\star}\beta^2_{\star} - 1)(1+\eta) + \frac{7}{4}\Gamma_{\star}\beta_{\star} = 0$$
(18)

For the highly relativistic upstream motion, $\Gamma_u \to \infty$ this equation yields, as expected, $\beta_{\star} \to \frac{1}{\sqrt{3}}$. Fig. (5) depicts the dependence of the flow velocity at the sonic point on the upstream velocity.

There is no steady state solution with $Q > Q_{\text{max}}$. We expect that if this value is reached than either the downstream readjust itself and the emissivity is reduced or a non steady state turbulent solution develops.

4 B. Skuratovsky et al.

It is interesting to note that this relativistic solution is equivalent to the classical Rayleigh flow (e.g. Emanuel 1986) that describes the evolution of the flow when heat is added or removed from a compressible ideal Newtonian fluid in a constant area duct. The flow chokes thermally if too much energy is added. This happens at a sonic point that corresponds to the critical value of heat deposition. In our setting the modified upstream is equivalent to a supersonic Rayleigh flow with heat deposition.

3 INSTABILITY OF THE CONTINUOUS SOLUTION

In the previous section, we found a steady-state solution only if the fraction of the energy transferred from the downstream to the upstream flow does not exceed the value given by Eq. (14). The upstream velocity, and therefore the shock discontinuity, decreases when the absorbed energy increases. At $Q = Q_{\text{max}}$, the flow velocity reaches the sound speed at x = 0 so that the transition to the subsonic downstream flow becomes continuous. The absence of steady-state solutions at $Q > Q_{\text{max}}$ implies that at this condition, the flow becomes unsteady. This conjecture is supported by the fact that the continuous solution at $Q = Q_{\text{max}}$ is in fact unstable: the amplitude of small sound perturbations diverges when the flow approaches the sound point. The reason is that the sound perturbation is dragged forward by the flow. Therefore the velocity of perturbations, that are directed backward in the flow frame, drops to zero at the sound point, which implies that the amplitude diverges. In this section, we present the formal derivation of this instability.

For simplicity, we assume that $\Gamma_u \gg 1$. In this case, most of the energy is converted into the thermal energy of leptons in the vicinity of the sound point where the flow is only mildly relativistic. Therefore one can neglect the contribution of cold protons into the equation of state and write w = 4p. In this case, the flow is described only by the energy and momentum equations (2), whereas using the continuity equation is not necessary. We project the equations of motion (2) on the 4-velocity and on the perpendicular direction. For this purpose, make a dot product of the equations with u_v and with the projection operator $g_{\mu\nu} - u_{\mu}u_{\nu}$. This yields

$$\frac{\partial w u^{\mu}}{\partial x^{\mu}} - u^{\mu} \frac{\partial p}{\partial x^{\mu}} = \frac{dS^{\mu}}{dx^{1}} u_{\mu}$$
(19)

$$wu^{\mu}\frac{\partial u_{\nu}}{\partial x^{\mu}} - \frac{\partial p}{\partial x^{\nu}} + u_{\nu}u^{\mu}\frac{\partial p}{\partial x^{\mu}} = \frac{dS_{\nu}}{dx^{1}} - u_{\nu}u_{\mu}\frac{dS^{\mu}}{dx^{1}}$$
(20)

We assume for simplicity that the momentum source is much smaller than the energy source so that $S^1 = 0$. Then we get for a one-dimensional flow

$$4\left(\frac{\partial p\Gamma}{\partial t} + \frac{\partial p\Gamma\beta}{\partial x}\right) - \Gamma\frac{\partial p}{\partial t} - \Gamma\beta\frac{\partial p}{\partial x} = \Gamma\frac{\partial S^{0}}{\partial x}$$
(21)

$$4p\left(\frac{\partial\Gamma\beta}{\partial t} + \beta\frac{\partial\Gamma\beta}{\partial x}\right) + \Gamma\beta\frac{\partial p}{\partial t} + \Gamma\frac{\partial p}{\partial x} = -\Gamma\beta\frac{\partial S^{0}}{\partial x}.$$
 (22)

In the steady state, these equations are reduced to

$$4\bar{p}\bar{\Gamma}^{2}\bar{\beta}' + 3\bar{\beta}\bar{p}' = S'^{0};$$
(23)

$$4\bar{p}\bar{\Gamma}\bar{\beta}^2\bar{\beta}'+\bar{p}'=-\bar{\beta}S'^0\,,\tag{24}$$

where prime denotes a *x* derivative, and we used bar to denote parameters in the steady state. Eliminating \bar{p}' from these equations yields

$$4\bar{p}(1-3\bar{\beta}^2)\bar{\Gamma}^2\bar{\beta}' = (1+3\bar{\beta}^2)S'^0,$$
(25)

which means that the flow passes the sonic point, $\beta = 1/\sqrt{3}$, only

if S'_e vanishes at this point, consistent with our findings about the critical point.

To study the stability of the steady-state flow, we consider a small perturbation to the steady state solution, $\beta = \overline{\beta} + \delta\beta$ and $p = \overline{p} + \delta p$, where $\delta\beta \ll \overline{\beta}$; $\delta p \ll \overline{p}$. Linearizing the equations, we get:

$$3\left(\frac{\partial\delta p}{\partial t} + \bar{\beta}\frac{\partial\delta p}{\partial x}\right) + 4\bar{p}\bar{\Gamma}^{2}\left(\bar{\beta}\frac{\partial\delta\beta}{\partial t} + \frac{\partial\delta\beta}{\partial x}\right) + 4\bar{\beta}'\bar{\Gamma}^{2}\delta p + (3\bar{p}' + 8\bar{p}\bar{\beta}\bar{\beta}'\bar{\Gamma}^{4})\delta\beta = 0, \qquad (26)$$

$$\bar{\beta}\frac{\partial\delta p}{\partial t} + \frac{\partial\delta p}{\partial x} + 4\bar{p}\bar{\Gamma}^{2}\left(\frac{\partial\delta\beta}{\partial t} + \bar{\beta}\frac{\partial\delta\beta}{\partial x}\right) + 4\bar{\beta}'\left[\bar{\beta}\bar{\Gamma}^{2}\delta p + \bar{p}\bar{\Gamma}^{4}(1 + \bar{\beta}^{2})\delta\beta\right] = -\delta\beta S'^{0}. \qquad (27)$$

Assuming high frequency oscillations, such that c/ω is much smaller than the characteristic scale of the flow, we employ the WKB approximation. For the non-relativistic Rayleigh flow, the procedure has been developed by Umurhan (1999).

Expanding the solution in powers of $1/\omega$ we obtain:

$$\begin{pmatrix} \delta\beta\\ \deltap \end{pmatrix} = \begin{pmatrix} \beta_0 + \beta_1/\omega + \dots\\ p_0 + p_1/\omega + \dots \end{pmatrix} \exp\left[i\omega\left(-t + \int \frac{dx}{u}\right)\right],$$
(28)

where *u* is the wave phase velocity. Substituting this ansatz into Eqs. (26) and (27) and collecting the terms of the order of ω one gets

$$3\left(-1+\frac{\bar{\beta}}{u}\right)p_0+4\bar{p}\bar{\Gamma}^2\left(-\bar{\beta}+\frac{1}{u}\right)\beta_0=0, \qquad (29)$$

$$\left(-\bar{\beta} + \frac{1}{u}\right)p_0 + 4\bar{p}\bar{\Gamma}^2\left(-1 + \frac{\bar{\beta}}{u}\right)\beta_0 = 0.$$
(30)

This set of equations has a nontrivial solution if

$$u = \frac{\bar{\beta} \pm 1/\sqrt{3}}{1 \pm \bar{\beta}/\sqrt{3}} \,. \tag{31}$$

We see that the wave velocity is a relativistic superposition of the flow velocity, $\overline{\beta}$, and the sound velocity, $\pm 1/\sqrt{3}$. Substituting *u* back to Eq. (30) we find a relation between p_0 and β_0 :

$$p_0 = \pm \frac{4}{\sqrt{3}} \bar{p} \bar{\Gamma}^2 \beta_0 .$$
 (32)

In the next approximation, one collects terms of the order of ω^0 . This yields a set of equations of the form

$$i\hat{\mathbf{M}} \begin{pmatrix} \beta_1 \\ p_1 \end{pmatrix} = \mathbf{\Phi} = -$$
(33)
$$\begin{pmatrix} 3\bar{\beta}p'_0 + 4\bar{\rho}\bar{\Gamma}^2\beta'_0 + 4\bar{\beta}'\bar{\Gamma}^2p_0 + (3\bar{\rho}' + 8\bar{\rho}\bar{\beta}\bar{\beta}'\bar{\Gamma}^4)\beta_0 \\ p'_0 + 4\bar{\rho}\bar{\beta}\bar{\Gamma}^2\beta'_0 + 4\bar{\beta}' \left[\bar{\beta}\bar{\Gamma}^2p_0 + \bar{\rho}(1+\bar{\beta}^2)\bar{\Gamma}^4\beta_0\right] + \beta_0 S'^0 \end{pmatrix},$$

where the matrix $\hat{\mathbf{M}}$ is the same as in Eqs. (29,30). Substituting *u* from Eq. (31), we find

$$\hat{\mathbf{M}} = \frac{1}{\bar{\Gamma^2}(\bar{\beta} \pm 1/\sqrt{3})} \begin{pmatrix} \mp\sqrt{3} & 4\bar{\rho}\bar{\Gamma^2} \\ 1 & \mp\frac{4}{\sqrt{3}}\bar{\rho}\bar{\Gamma^2} \end{pmatrix}.$$
 (34)

Substituting \bar{p}' from (24), p_0 from (32) and S'_e from (25) into Φ yields:

$$\boldsymbol{\Phi} = -4\bar{p}\bar{\Gamma}^{2} \begin{pmatrix} (1 \pm \sqrt{3}\bar{\beta})\beta_{0}' + \bar{\beta}' \left\{ \frac{2\left(\bar{\beta} \pm \frac{2}{\sqrt{3}}\right)}{1-\bar{\beta}^{2}} - \frac{2\left(\bar{\beta} \pm \sqrt{3}\bar{\beta}^{2}\right)}{\bar{\beta}^{2} + \frac{1}{3}} \right\} \beta_{0} \\ (\bar{\beta} \pm \sqrt{3})\beta_{0}' + \bar{\beta}' \left\{ \frac{\bar{\beta}^{2} \pm \frac{4}{\sqrt{3}}\bar{\beta} + 1}{1-\bar{\beta}^{2}} - \frac{\bar{\beta}^{2} \pm \frac{2}{\sqrt{3}}\bar{\beta} - \frac{1}{3}}{\bar{\beta}^{2} + \frac{1}{3}} \right\} \beta_{0} \end{pmatrix}$$
(35)

MNRAS 000, 1-?? (2023)

$$\operatorname{Det}\begin{pmatrix} \mp\sqrt{3} & -(1\pm\sqrt{3}\bar{\beta})\beta_{0}'+\bar{\beta}'\beta_{0}\left\{\frac{2(\bar{\beta}\pm2/\sqrt{3})}{1-\bar{\beta}^{2}}-\frac{2(\bar{\beta}\pm\sqrt{3}\bar{\beta}^{2})}{\bar{\beta}^{2}+1/3}\right\}\\ 1 & -(\bar{\beta}\pm\sqrt{3})\beta_{0}'-\bar{\beta}'\beta_{0}\left\{\frac{2(\bar{\beta}\pm2/\sqrt{3})}{1-\bar{\beta}^{2}}-\frac{2(\bar{\beta}\pm\sqrt{3}\bar{\beta}^{2})}{\bar{\beta}^{2}+1/3}\right\} \end{pmatrix} = 0$$
(36)

Choosing the lower sign of \pm in (36), which describes the wave propagating backwards in the comoving frame, leads to

$$\beta_{0}'(\bar{\beta} - \frac{1}{\sqrt{3}}) = -\bar{\beta}'\beta_{0} \left\{ \frac{\frac{\bar{\beta}^{2}}{2} - \sqrt{3}\bar{\beta} + \frac{7}{6}}{1 - \bar{\beta}^{2}} - \frac{\frac{3}{2}\bar{\beta}^{2} - \frac{2}{\sqrt{3}}\bar{\beta} - \frac{1}{6}}{\bar{\beta}^{2} + \frac{1}{3}} \right\}$$
(37)

Integrating (37) yields

$$\beta_0 = \frac{C}{\bar{\Gamma}} \cdot \frac{\bar{\beta}^2 + \frac{1}{3}}{\bar{\beta} - \frac{1}{\sqrt{3}}} \cdot \left(\frac{1 + \bar{\beta}}{1 - \bar{\beta}}\right)^{\frac{1}{\sqrt{3}}},\tag{38}$$

where C is an integration constant. Since p_0 is proportional to β_0 by Eq. (32), we conclude that at the zeroth order, both $\delta\beta$ and δp diverge as $\bar{\beta}$ approaches the speed of sound from above, meaning that transition through the critical point is unstable. The result can be understood physically. Perturbations in the upstream propagate at the speed of sound both towards and away from the shock. As long as the shock speed is supersonic, perturbations from the upstream won't reach the shock. However, as we reach the critical energy transfer, the shock speed approaches the speed of sound. At this stage, perturbations that move towards the shock reach it and accumulate there, resulting in overall instability.

4 DISCUSSION

We presented a common framework unifying the description of macroscopic precursor phenomena in relativistic shock waves. Our general results for the perturbed upstream structure agree nicely with those obtained in numerical PIC simulationa (e.g. Lemoine et al. 2019) in which high energy particles accelerated in the downstream deposit their energy in the upstream. In a steady state, there is an upper limit to the fraction of energy transferred from the downstream to the upstream (Derishev & Piran 2016). This maximal energy decreases when the momentum transferred increases. It approaches a constant value, $(2 - \sqrt{3})/(\sqrt{3} + 2\eta) \approx 0.155/(1 + 1.15\eta)$, in the extreme relativistic case, when the Lorentz factor of the cold upstream approaches infinity.

There are several implications to this result. The first corollary deals with radiation-mediated shocks, in which energy transport from the downstream to the upstream is a dominant phenomenon. Typically, relativistic radiation mediated shocks involve a collisionless subshock. Our analysis explains this phenomenon. The radiation flux from the downstream may not be fine-tuned to the value required for a smooth transition. It is important to stress that for radiation-mediated shocks our solution is valid only for the upstream region. In this region any photon that interacts with the relativistic flow creates a pair that moves with the flow. Indeed, in this region our solution is consistent with those of Budnik et al. (2010a) and Granot et al. (2018). However, the downstream is mildly relativistic and there the photons' behavior is diffusive. For this reason Fig. 2 cannot be used to estimate the subshock jump in these shocks. Moreover, the stability analysis discussed in §3 may not be applicable.

A second novel result is that once the energy transport to the

precursor reaches a maximal value, the shock disappears, and the solution passes through a sonic point. However, this marginal solution is unstable. The physical reason for the instability is apparent. Perturbations in the upstream move at the speed of sound relative to the local flow. When the upstream velocity ahead of the shock approaches the speed of light, the "outgoing" perturbation directed towards the far upstream is carried towards the sonic point. As they cannot cross the sonic point, they accumulate there, resulting in instability.

This instability implies that if, for some reason, the energy flux from the hot downstream is too large, it will induce unstable turbulence in the upstream. The characteristic scale of this turbulence will be macroscopic, as it should correspond to the scale in which the energy flux is deposited in the upstream. One can speculate that such large-scale turbulence could be the source of large-scale magnetic fields. If correct, this will resolve one of the puzzles involving collisionless shocks in GRB afterglows - what is the origin of the large scale (much larger than the local plasma skin depth) magnetic fields that are implied from the afterglow observations (Gruzinov & Waxman 1999).

ACKNOWLEDGEMENTS

We thank Evgeny Derishev, Ehud Nakar and Re'em Sari for helpful discussions. This work was supported by an Advanced ERC grant TReX and by ISF grants 2126/22 (TP) and 2067/19 (YL).

DATA AVAILABILITY

No new data has been generated.

REFERENCES

- Bell A. R., 2004, MNRAS, 353, 550
- Budnik R., Katz B., Sagiv A., Waxman E., 2010a, The Astrophysical Journal, 725, 63
- Budnik R., Katz B., Sagiv A., Waxman E., 2010b, ApJ, 725, 63
- Couch S. M., Milosavljević M., Nakar E., 2008, ApJ, 688, 462
- Derishev E. V., Piran T., 2016, Mon. Not. R. Astron. Soc., 460, 2036
- Emanuel G., 1986, Gasdynamics. American Institute of Aeronautics and Astronautics, New York, doi:10.2514/4.861871, https://arc.aiaa. org/doi/abs/10.2514/4.861871
- Garasev M., Derishev E., 2016, MNRAS, 461, 641
- Granot A., Nakar E., Levinson A., 2018, MNRAS, 476, 5453
- Gruzinov A., Waxman E., 1999, ApJ, 511, 852
- Hoshino M., 2008, ApJ, 672, 940
- Iwamoto M., Amano T., Matsumoto Y., Matsukiyo S., Hoshino M., 2022, ApJ, 924, 108
- Lemoine M., Vanthieghem A., Pelletier G., Gremillet L., 2019, Physical Review E, 100, 033209
- Lyubarsky Y., 2006, ApJ, 652, 1297
- Lyubarsky Y., 2018, MNRAS, 474, 1135
- Nakar E., Sari R., 2012, ApJ, 747, 88
- Nakar E., Bret A., Milosavljević M., 2011, ApJ, 738, 93
- Sironi L., Spitkovsky A., 2011, ApJ, 726, 75
- Umurhan O. M., 1999, Annual Research Briefs, Center for Turbulence Research (1999):, pp 99–108